COMPARISON OF LOW- AND HIGH TEMPERATURE DIFFERENTIAL STIRLING ENGINES

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ABSTRACT

Stirling engines can be operated with various heat sources. We have developed several Stirling engines, included a low temperature differential Stirling engine using hot spring heat and a high temperature differential Stirling engine using a combustion gas as the fuel. These engines have different features and characteristics, and have different thermal performance.

In this paper, the performances of these difference types of Stirling engines were measured, were compared using the normalized values.

INTRODUCTION

In the field of the low temperature differential Stirling engines, several small Stirling engines have developed by I. Kolin, J. R. Senft and so $on^{1),2}$. We have being developed the low temperature differential Stirling engine with several hundreds wattage of an output power³⁾. The engine uses hot spring heat or hot water from industrial waste heat as the fuel.

In this paper, we measured heat losses and powers of the engine in order to estimate the engine performance. Then the performance of a prototype of the low temperature differential Stirling engine (LDSE) was compared with that of a high temperature differential Stirling engine (HDSE) after their experimental results had been normalized.



Figure 1 Feature of LDSE

STRUCTURE OF LDSE

Figure 1 and Fig. 2 show the features and structure of LDSE. Table 1 shows specifications of this engine and HDSE⁵⁾ used to compare the performances.

LDSE uses a gamma-type configuration, because this type engine can keep a large heat transfer area of heat exchangers. The crank mechanism is covered with a buffer tank, and the engine can be pressurized above atmospheric pressure.

The heat exchanger is located around the displacer cylinder as shown in Fig. 2. The heater

and the cooler are fin-tube type heat exchangers. The hot water flows in the heater tubes. The cooling water flows in the cooler tubes. The working gas flows out of these tubes.

The crank mechanism of the Compression piston rotated space power is а Scotch-Yoke mechanism. This mechanism makes the crank assembly compact and decreases the side force on the power piston. The crank mechanism of the displacer piston uses a normal type mechanism. crank These don't mechanisms use any lubricating devices like an oil pump. The lubricant for the mechanism is only grease in the bearings. An one-way-type piston ring, which being developed has in our laboratory is used as the seal device on the power $piston^{4}$.

It has the same effect as a pump, and a mean pressure in the working space is kept higher than the buffer space pressure. It is a suitable seal device for the low temperature difference Stirling engines, because it has low leakage of the working gas and a low friction loss. Two one-way-type rod seals, which have the same construction are used as the



Table 1 Specifications				
	Low temperature difference Stirling engine (LDSE)	High temperature difference Stirling engine (HDSE)		
Enginer type	Gamma configuration	Gamma configuration		
Power piston				
Swept volume	25100 cm ³	81 cm ³		
Bore / Stroke	400 mm / 200 mm	72 mm / 20mm		
Displacer piston				
Swept volume	40200 cm ³	81 cm ³		
Bore / Stroke	800 mm / 80 mm	72 mm / 20mm		
Phase angle	90 deg	eg 90 deg		
Heater type	Fin tube type	Moving tube type		
Regenerator				
Туре	Stacked mesh	Spring Mesh / Stacked mesh		
Material	Copper	Stainless steel		
Cooler type	Fin tubetype	Moving tube type		
Crank mechanism	Scotch-yoke	Scotch-yoke		
Heat source	Hot water (Hot spring)	Combustion gas		
Working gas	Air	Helium / Air		
Mean engine pressure	0.1 MPa ~ 0.3 MPa	~ 1 MPa		
Expansion space temp.	~ 403 K	~ 9 <mark>23 K</mark>		
Compression space temp.	~ 313 K (Water cooling)	~ 343 K (Water/Air cooling)		

Cracifications

Tabla 1

the same construction are used as the seal devices on the displacer rods.

STRUCTURE OF HDSE

Figure 3 shows the structure of HDSE, which was used in the following experiments. The displacer piston of this engine has both heating and cooling inner tubes for the working gas

which flows to and from the outer tubes. A generator is located in the displacer piston.

A Scotch-Yoke mechanism is used as the crank mechanism. This keeps the crank mechanism compact. A 4-piece-type piston ring is used as the seal device of each piston. A mechanical seal is used as the seal device between the buffer space and the atmosphere.

ESTIMATED METHODS

Figure 4 shows a heat balance, which is discussed in this paper. This figure shows that an overall efficiency of the Stirling engine is calculated in Eq. 1 using a burner efficiency,



Figure 3 Schematic view of HDSE

(1)

(2)

 $\eta_{\rm b}$ (=Q_{in}/Q_{total}), an indicated efficiency, $\eta_{\rm i}$ (=L_i/Q_{in}) and a mechanical efficiency, $\eta_{\rm m}$ (=L_s/L_i).

 $\eta_{total} = \eta_b \times \eta_i \times \eta_m$

By the rough considerations, the burner efficiency is affected by the shape of the heat source, the indicated efficiency is affected by the heat flow in the engine, and the mechanical efficiency is affected by the mechanism and the operating conditions. The burner efficiency is difficult to consider because the engines used for this testing have different shaped heat sources. The following heats and efficiencies are based on the heat input, Q_{in}.

HEATS AND LOSSES

The heat input, Q_{in} is the overall heat that enters the engine. The heat input, Q_{in} is calculated by Eq. 2 using indicated power, L_i (measured using P-V diagrams) and rejected heat, Q_{reject} as described later.

$$Q_{in} = L_i + Q_{reject}$$

The expansion space power, L_E is calculated by Eq. 3 using an expansion space pressure,

 P_E , a volume change of the expansion space, dV_E and the engine speed, n (rps). The expansion space power, L_E is measured using the overall heat that flows from the heat source to the working gas through the heater.

$$L_{E} = \oint P_{E} dV_{E} \cdot n \tag{3}$$

The compression space power, L_C is calculated by Eq. 4 using the compression space pressure, P_C and a volume change of the



compression space, dV_C.

$$L_{\rm C} = -\oint P_{\rm C} dV_{\rm C} \cdot \mathbf{n} \tag{4}$$

The indicated power, L_i is determined by the difference between L_E and L_C as shown in Eq. 5.

$$L_{i} = \oint P_{E} dV_{E} + \oint P_{C} dV_{C} = L_{E} - L_{C}$$
(5)

The output power, L_s is calculated by Eq. 6 using break torque, T_q which was measured in the experiments.

$$\mathbf{L}_{s} = 2\pi \cdot \mathbf{I}_{q} \cdot \mathbf{n} \tag{6}$$

The rejected heat, Q_{reject} is calculated by Eq. 7 using a specific heat of the cooling water, c (J/kgK), an inlet water temperature, T_{in} , an outlet water temperature, T_{out} and a water mass flow, m_{water} (kg/s).

$$Q_{\text{reject}} = c \cdot m_{\text{water}} (T_{\text{out}} - T_{\text{in}})$$
(7)

Other thermal loss, Q_{other} is calculated by Eq. 8. It is considered that this loss consists mainly of a heat conduction loss and a regenerator loss. $Q_{++} = Q_{+-} - I_{+} = Q_{++} + I_{+}$ (8)

$$Q_{other} = Q_{in} - L_e = Q_{reject} + L_c$$
(6)
A pressure loss, L_{ploss} is defined by Eq. 9.

$$L_{ploss} = -\oint (P_e - P_c) dV_e \cdot n$$
(9)

A mechanical loss, L_m is the difference between L_i and L_s as shown in Eq. 10. $L_m = L_i - L_s$ (10)

NORMALIZED SPEED AND HEATS

In order to compare different types of Stirling engines, the engine speed and each heats should be normalized. The performances of LDSE and HDSE can be estimated directly using the normalized values.

The engine speed band is different between that of LDSE and HDSE under all operating conditions. A normalized speed, N^* is defined by Eq. 11 using the engine speed, N and a rated engine speed, N_{rate}.

$$N^* = \frac{N}{N_{rate}}$$
(11)

In the case of HDSE, the rated engine speed is set to 1000 rpm. This value is the rated speed of this design. This engine speed, 1000 rpm is the maximum permitted speed with continued safe operation. The maximum permitted speed of LDSE is higher than its power output band. In the case of LDSE, the rated engine speed, N_{rate} is assumed to be set to the engine speed when it operates at the maximum output power.

In the case of the heat and losses, Q, each normalized heat, Q^* is defined by Eq. 12.

$$Q^* = \frac{Q}{Q_{in}}$$
(12)

EFFICIENCIES

The indicated efficiency, η_i , the mechanical efficiency, η_m and an internal conversion

efficiency, η_{int} are defined by following equations.

$$\eta_{i} = \frac{L_{i}}{Q_{in}}$$

$$\eta_{m} = \frac{L_{s}}{L_{i}}$$

$$\eta_{int} = \frac{L_{i}}{L_{e}}$$
(13)
(14)
(15)

EXPERIMENTAL METHODS OF LDSE

Table 2 shows experimental conditions of LDSE. The buffer pressure was maintained at atmospheric pressure (0.1 MPa) in this experiment. Ethylene glycol, maintained at a temperature of 403 K and the

Table 2	Experimental	conditions	of LDSE
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Heat source	Ethylene glycol
Temp. of heat source	403 K
Working gas	Air
Buffer pressure	0.1MPa (atmosphere)
Regenerator	Stacked Mesh (#24 × 30 seats)
Cooling type	Water cooling
Piston ring	One-way-type

volume flow to 20 l/min was used as the heat source. The compression space pressure and a pressure difference between the expansion and the compression space were measured by strain gauge type sensors. Several thermocouples were used to measure temperatures. The expansion gas temperature, T_E and the compression space temperature, T_C were measured as representative gas temperatures at points shown in Fig. 2. The load to the engine output was adjusted by a rope break, and the engine torque, T_q was measured by spring scales which were located at both ends of the rope.

EXPERIMENTAL METHODS OF HDSE

Table 3 shows experimental conditions of HDSE. A mean pressure of the working space was set to 0.8 MPa. A 1 kW (max.) electric heater was used to heat the engine. Heat input from the electric heater was adjusted to maintain the heater wall

 Table 3
 Experimental condition of HDSE

Heat source	Electric heater
Heater wall temp.	973 K
Working gas	He
Mean engine pressure	0.8 MPa
Regenerator	Spring Mesh
Cooling type	Water cooling
Piston ring	4-piece-type

temperature, T_W shown in Fig. 3 to 973 K. T_w indicates the highest temperature in the engine system. The expansion temperature, T_E and the compression temperature T_C were measured at points shown in Fig.3 using thermocouples. The load to the engine output was adjusted by changing a value of a resistance which was connected to a generator. The arm attached to the generator was connected to an electric balance to measure the engine torque, T_q .

EXPERIMENTAL RESULTS

ENGINE PERFORMANCE OF LDSE

Figure 5 shows the relation between the engine speed and the indicated and output power



of LDSE. This figure shows that the maximum output power reached to 146 W at the engine speed, 143 rpm. The maximum speed reached 247 rpm at no load.

Figure 6 shows the relation between the engine speed and the expansion and compression temperatures, T_E and T_C . These values of the temperatures changed somewhat by changing the engine speed. T_E was 381 K and T_C was 313 K at the rated engine speed, 143 rpm.

Figure 7 shows the relation between the engine speed and the heat input, Q_{in} , which is determined in Eq. 2. From this figure, the heat input increases proportionally with higher engine speed.

ENGINE PERFORMANCE OF HDSE

Figure 8 shows the relation between the engine speed and the indicated and output power of HDSE. In the HDSE, the maximum points of the indicated and output power cannot be seen. In this experiment, the gas temperatures were not affected by the engine speed hardly, and T_E was about 733 K and T_C was about 309 K.

Figure 9 shows the relation between the engine speed and the heat input, Q_{in} , which is led to Eq. 2. From this figure, the heat input increases proportionally with higher engine speed too.



Figure 7 Heat input as a function of engine speed (LDSE)



NORMALIZED HEATS

Figure 10 shows the relation between the normalized speed, N^* and the normalized expansion and compression power, L_E^* , L_C^* . From this figure, L_E^* of LDSE converged to that of HDSE. It means that the heat input flows to the working gas with the same rates. On the other hand, L_C^* of LDSE was higher than that of HDSE. It means that the heat input of LDSE is used in order to cool the working gas with a larger rate than that of HDSE.

Figure 11 shows the relation between the normalized speed, N^* and the normalized rejected and other heats, Q_{reject}^* and Q_{other}^* . From



this figure, Q_{reject}^* of LDSE is higher than that of HDSE, and reaches 0.95 at the rated engine speed. Q_{reject}^* of LDSE increases with higher engine speed, though that of HDSE decreases with higher engine speed. It means that LDSE doesn't use the heat input effectively at the high engine speed band. And a next subject of LDSE is how it uses the heat input effectively.

 Q_{other} of LDSE converged to that of HDSE, and their values are about 0.5. As described above, Q_{other} is mainly the heat conduction loss and the regenerator loss, but accounting for these losses and these heats in this experiment was difficult. A detailed accounting of these losses is needed.

Figure 12 shows the relation between the normalized speed, N^* and the normalized pressure loss, L_{ploss}^* . In this figure, L_{ploss}^* of LDSE is lower than that of HDSE somewhat. The pressure loss and the mechanical loss in the LDSE determine the engine speed band. A way to decrease these losses should be considered.



normalized speed





Figure 13 shows the relation between the normalized speed, N^* and the normalized mechanical loss, L_m^* . In this figure, L_m^* of LDSE is much lower than that of HDSE. L_m^* of HDSE increases with higher engine speed, but that of LDSE keeps constant value without the effect of the engine speed. This is one of the characteristics of LDSE, which operates in the low engine speed band.

EFFICIENCIES

Figure 14 shows the relation between the normalized speed, N^* and the indicated efficiency, η_i . This figure shows that η_i of LDSE is 0.05 and that of HDSE is 0.2 at the rated engine speeds. The difference between that of LDSE and HDSE is explained clearly with a consideration of a theoretical efficiency (Carnot efficiency) which is determined by the gas temperature ratio. The indicated efficiency can be increased by decreasing the other thermal losses, Q_{other} .

Figure 15 shows the relation between the normalized speed, N^* and the mechanical efficiency, η_m . It is difficult to compare these experimental results directly, because these engines are designed for experimental use. The mechanical efficiency, η_m of LDSE can be increased easily, because it operates at a low





engine speed. It is one of the good characteristics of LDSE.

Figure 16 shows the relation between the normalized speed, N^* and the internal conversion efficiency, η_{int} . The value of this efficiency is affected by that of the Carnot efficiency. The Carnot efficiency (=1-T_C/T_E) of LDSE is calculated to 0.20 using the expansion and compression space gas temperatures, T_E and T_C at the rated engine speed. That of HDSE is calculated to 0.58 with the same condition. This figure shows that of LDSE is about 50% of the Carnot efficiency at the rated engine speed. The internal conversion efficiency, η_{int} of LDSE can be increased by the decrease of the pressure loss.

CONCLUSION

The difference types of Stirling engines are compared. The summary of the experimental and estimated results is shown below.

- (1) LDSE reached 145 W of output power at the engine speed, 142 rpm and the buffer pressure, 0.1 MPa using 403 K of the heat source temperature.
- (2) The indicated efficiency of LDSE reached to 0.05, when it operated at the maximum output power.
- (3) The heat input of LDSE is used in order to cool the working gas with a larger rate than that of HDSE. And the normalized rejected heat of LDSE was higher than that of HDSE. A subject for further research on LDSE is how it uses the heat input effectively.

The Carnot thermal efficiency of the low temperature differential Stirling engine will not reach that of the high temperature differential Stirling engine. But the low temperature differential Stirling engine can use many different kinds of heat sources effectively. It needs to be developed more. We will have developed it using the above experimental results.

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